

Technology Considerations for Advanced Formation Flight Systems

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How Can Technologies Impact System Concept

- Need (Technology Pull)
 - □ Technologies can fulfill need or requirement
 - □ Technologies can overcome barriers (limitations, constraints, etc.)

• Opportunity (Technology Push)

- □ Technologies can Create Opportunities
- □ New Capabilities
- □ Competitive advantage
 - ♦ Cost
 - Performance
 - ♦ Maintenance
 - ♦ Other



Formation System Concept is Itself a Technology

• Needs

- □ Efficient Transport
 - ◆ Fuel
 - ♦ Cost
 - ♦ Crew, Maintenance…
 - Operational Access (Noise, Runways)
 - ♦ Flexibility
 - ♦ Others

Opportunity

- □ Different design space if use multiple vehicles
- □ Overcome constraints (eg runway width, single departure point)
- □ Performance
 - ◆ Fuel efficiency, crew
- Development of key technologies enable formation flight
- □ Flexibility
- □ Runway Throughput



What are the Key Technologies for Formation Flight

- Start with Fundamental Abstraction of System or Concept (many ways)
 - □ Functional
 - □ Operational
 - Concept of Operations
 - □ Physical
 - □ Component
 - □ Constraint
 - □ Information

Based on Abstract view, identify

- ☐ Technology needs☐ Key questions
- D Potential opportunities

• Useful to sketch elements to visualize system

□ Multiple views















What are the Key Technologies for Formation Flight

Overall Concept Questions

- □ Concept of Operations?
- $\hfill\square$ How does form up occur
- □ Station keeping requirements
- □ Failure Modes
- □ Existing elements or New
 - Vehicles
 - Control Systems
 - CNS
 - ♦ Other



- Concept Scale Opportunities/Costs
 - □ Performance gains estimate
 - ♦ Fuel
 - Capacity
 - □ Costs
 - Development
 - Deployment

- Concept Technologies Reqs
 - □ Formation design
 - □ Station Keeping
 - ♦ Com
 - Nav
 - ♦ Surveillance
 - Control







What are the Key Technologies for Formation Flight



- Communications
- Navigation
- Surveillance
- Control (Station Keeping)
 - Intent States
 - □ String Stability
- Vehicle Configuration
 - □ Aero/Performance □ Control
- Propulsion
- Degree of Autonomy
- Flight Criticality
 - □ Hardware
 - □ Software
- Low Observability
- Others?





Communications

• Requirements

- □ Communicate necessary information between formation elements and command node (LAN and Air-Ground)
- □ Bandwidth
- □ Low-Observable?
- □ Synchronous vs asynchronous

Constraints

- □ Spectrum
- □ Antenna Location

Technologies

- □ Radio
 - ♦ UHF, VHF, MMW
- □ Optical
 - ♦ Laser
- □ Protocols



COMMUNICATION

- Voice
 - □ VHF (line of sight)
 - ◆ 118.0-135.0 Mhz
 - .025 spacing in US, 0.083 spacing in Europe)
 - - ◆ 230-400 Mhz (guess)
 - \Box HF (over the horizon)
 - □ Optical (secure)

Datalink

- □ ACARS (VHF) VDL Mode 2
- □ VDL Modes 3 and 4 (split voice and data)
- □ HF Datalink (China and Selcal)

Geosynchronous (Inmarsatt)

- Antenna Requirements
- LEO and MEO Networks
- Software Radios
- Antenna Requirements



Generic Avionic System





Navigation (relates to Surveillance)

Requirements

- □ General Navigation (medium precision)
- □ Station Keeping (high precision)
- □ Integrity
- □ Availability

Constraints

Existing nav systemsLoss of signal

Technologies

□ GPS/Galileo (need Differential)

Code vs Carrier Phase Approaches

□ IRS/GPS

□ Sensor Based Approaches for Station Keeping

- ♦ Image (Visible, IR)
- ♦ Range Finders (Laser, Ultrasonic)





XYZT The Global Positioning System Measurements of code-phase arrival times from at least four satellites are used to estimate four quantities: position in three dimensions (X, Y, Z) and GPS time (T). P.H. Dana 5/10/98

(Courtesy of Peter Dana. Used with permission.)

From http://www.colorado.Edu/geography/gcraft/notes/gps/gps_f.html



Inertial Reference Unit

- Integrate acceleration from known position and velocity
 - □ Velocity
 - D Position

• Need Heading

- □ Gyros
 - Mechanical
 - ♦ Laser

• Can get Attitude

□ Artificial Horizon (PFD. HUD)

• Drift Errors

- □ IRU unusable in vertical direction (need baro alt)
- □ Inflight Correction
 - ◆ DME
 - ♦ GPS
 - Star Sighting for Space Vehicles
- Measurement Give Attitude Also
- 777 Analytical Redundancy



Surveillance

• Requirements

Observed states of lead elements sufficient to form-up and maintain station keeping either manually or by automatic control
 Feed forward states (intent)

Constraints

□ Sight Angles

□ Installation (weight, cost, power, etc)

□ Cooperative Targets

Technologies

- □ Automatic Dependant Surveillance Broadcast (ADS-B)
- □ Image Based Systems (Vis, IR)
- □ Radar (X Band, MMW0
- □ Range Finders (Laser)
- □ Sensor Fusion Systems







Bob Hilb UPS/Cargo Airline Association (Image removed due to copyright considerations.)





- Wavelength λ
 - \Box S Band (10 cm)
 - □ X Band (3 cm)
 - □ Ku Band (1 (cm)
 - □ Millimeter Wave (94 Ghz pass band)
- Radar Range Equation

• Beamwidth Θ

□ Θ = λ/D
□ D = Diameter of Circular Antenna
□ Pencil beam vs Fan Beam

Mechanically Steered Antennas

□ Scan and Tilt



INTENT REPRESENTATION IN ATC

Intent formalized in "Surveillance State Vector"



Accurately mimics intent communication & execution in ATC





RADAR SURVEILLANCE ENVIRONMENT

 Allows visualization of different (actual or hypothetical) surveillance environments

□ Useful for conformance monitoring analyses of impact of surveillance





ADS-B SURVEILLANCE ENVIRONMENT

- Potential access to more states (e.g. dynamic and intent)
- Need to assess benefits for conformance monitoring







- □ Maintain Station Keeping sufficient to achieve formation benefits
- □ Tolerance to Environmental Disturbances
- □ String stability

Constraints

- □ Certification
- □ Failure modes
- □ Available states

Technologies

- □ Performance seeking control
- □ Multi-Agent Control Architectures
- □ Distributed Control Approaches
- □ Leader-Follower Schemes
- □ Fault Tolerant Systems
 - Redundancy Architectures





 $\hfill\square$ Form up and station keeping may need to be automated

Constraints

- □ Reliability, integrity
- □ Certification
- □ Failure Modes

Technologies

- □ Flight Directors
- □ Autopilots
- □ Intercept systems





- □ High Integrity Implementation for Formation
- □ Formation requirement exceeds specs for current vehicles (eg 777)

Constraints

□ Failure Modes

Technologies

□ DO 178B □ ??



Aero-Configuration

• Requirements

- □ Mission based requirements (you will define)
- □ Formation based requirements
- □ Special Control Requirements

Constraints

- □ Stability and Control (CG)
- □ Formation and non-Formation operation

Technologies

□ Conventional approaches modified by formation considerations

- ♦ Asymmetric
- Formation optimal vs single optimal
 - ↓ Lead High WL, Low AR >> high vortex
- □ Vortex Tailoring
- □ Unique configurations or control systems





• Symmetric vs Asymmetric

• Variable

□ Formation vs Free Configurations

• Formation Specific Considerations

□ What is the optimal aspect ratio for overall performance

- Are there special, non-classical control needs?
- What are takeoff and landing considerations
- In-flight physical hookups





- □ Take-off, balanced field length >> drives thrust
- □ Cruise efficiency
- □ Response time

Constraints

□ Operational in formation and non formation configuration

Technologies

- □ Unmatched multi engines (shut down in cruise, eg Voyager)
- □ Broad operating envelope engines (SFC hit)
- □ Tow Schemes







Voyager aircraft return from non-stop trip around the world

Voyager

Formation Transport Example: ICAT C-47 (DC-3) towing CG-4 Cargo Gliders



Courtesy of the Atterbury-Bakalar Air Museum. Used with permission.

http://www.atterburybakalarairmuseum.org/CG4A_C47_color_photo.jpg



What are the risk considerations for technology incorporation

Readiness

□ NASA Technology Readiness Levels (TRL)

Vulnerability

- □ High (Key Element on Which Concept Based)
- □ Medium (Performance or Capability Enhancing, Competitive Factor)
- Low (alternatives available)

Competitive Risk

□ Goes both ways

Certification Risk

Operational Considerations

- □ Issues are discovered in field operations
 - Tracking Programs
- □ Unanticipated uses of technology



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